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Control of the PTO system of OWCs: feedback vs model predictive control

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Outline of the presentation

PART 1

- Generator feedback control: Mutriku test case
 - Power take-off system: biradial vs Wells turbine
 - Mathematical model
 - Generator feedback control law: computation and sensitivity analysis

PART 2

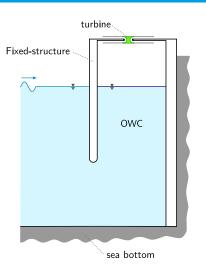
- Model predictive latching control: Spar-buoy OWC test case
- Power take-off system: biradial turbine equipped with an HSSV
- Changes to mathematical model
- A discrete control algorithm based on Pontryagin's Maximum Principle
- A new continuous control method based on the Discontinuous Galerkin Method
- Major conclusions

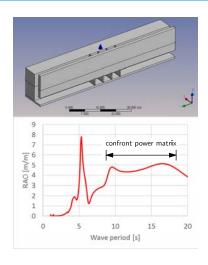


PART1 - Mutriku power plant test case



Hydrodynamic model

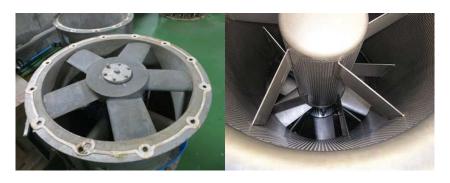




- Hydrodynamic model developed by Wanan Sheng
- High-order method, 2192 variables, 300 frequencies

Installed Mutriku power take-off

• Wells turbine with biplane rotor without guide vanes

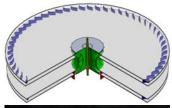


• Rotor diameter: 0.75 m. Generator rated power: 18.5 kW

Biradial turbine to be installed at Mutriku within OPERA H2020 Project

Biradial turbine with fixed guide vanes (Kymaner/IST patent WO/2011/102746)







- Rotor diameter: 0.50 m. Generator rated power: 30.0 kW
- To be installed at Mutriku in mid-April 2017 and in the Oceantec spar-buoy, deployed at BIMEP site, in Sept. 2017 (OPERA H2020 Project)

Turbine dimensionless curves

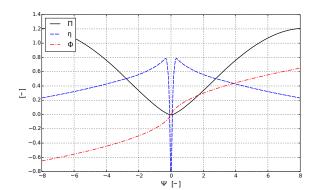
• The performance characteristics of a turbine in dimensionless form

$$\Psi = \frac{p}{\rho_{\mathsf{in}} \, \Omega^2 \, d^2}$$

$$\Phi = \frac{\dot{m}_{\text{turb}}}{\rho_{\text{in}} \Omega d^3}$$

$$\Pi = \frac{P_{\rm turb}}{\rho_{\rm in}\,\Omega^3\,d^5}$$

$$\eta = \frac{\Pi}{\Psi \Phi}$$



- d rotor diameter
- For Re $>10^6$ and Ma < 0.3 the dimensionless curves $\Phi,\,\Pi$ and η are independent of Re and Ma

$$\Rightarrow \dot{m}_{\text{turb}}(\mathbf{p}, \Omega, \rho_{\text{in}}) = \rho_{\text{in}} \Omega d^3 \Phi(\Psi), \qquad P_{\text{turb}}(\mathbf{p}, \Omega, \rho_{\text{in}}) = \rho_{\text{in}} \Omega^3 d^5 \Pi(\Psi)$$

$$P_{\text{turb}}(\boldsymbol{p}, \boldsymbol{\Omega}, \boldsymbol{\rho}_{\text{in}}) = \boldsymbol{\rho}_{\text{in}} \boldsymbol{\Omega}^3 d^5 \Pi(\boldsymbol{\Psi})$$



Mathematical model

• OWC motion is modelled as a rigid piston ($L_{\text{OWC}} \ll \lambda_{\text{waves}}$)

$$(\textit{m}_{2} + \textit{A}_{22}^{\infty}) \ \ddot{\textit{x}}_{2} = \underbrace{-\rho_{w}\textit{g}\textit{S}_{2} \, \textit{x}_{2}}_{\text{buoyancy}} \underbrace{-\textit{R}_{22}}_{\text{radiation diffraction}} \underbrace{+\textit{F}_{d2}}_{\text{air chamber}} \underbrace{-\textit{p}_{\text{atm}} \, \textit{S}_{2} \, \textit{p}^{*}}_{\text{air chamber}}$$

• Air chamber pressure (see [1, 2])

$$\dot{p}^* = -\gamma \left(p^* + 1\right) \frac{\dot{V}_c}{V_c} - \gamma \left(p^* + 1\right)^{(\gamma - 1)/\gamma} \frac{\dot{m}_{\mathsf{turb}}}{\rho \, V_c}$$

• Turbine/generator set dynamics

$$I\dot{\Omega}^* = \left(\,T_{
m turb} - T_{
m gen}\,
ight)\Omega_{
m max}^{-1} = \left(\,
ho_{
m in}\,\Omega^2 d^5\;\Pi(\Psi) - T_{
m gen}\,
ight)\Omega_{
m max}^{-1}$$

- Dimensionless variables
 - $oldsymbol{
 ho} p^* = rac{p-p_{
 m atm}}{p_{
 m atm}} \quad {
 m and} \quad \Omega^* = \Omega/\Omega_{
 m max}$
 - $\mathbb{O}[x_2] \approx \mathbb{O}[p^*] \approx \mathbb{O}[\Omega^*] \approx 1 \implies \text{Similar orders of magnitude for errors}$



Generator feedback control law

Generator feedback control law

• If the turbine operates at the best efficiency point η_{max}

$$P_{\mathsf{turb}} \approx \underbrace{\rho_{\mathsf{atm}} \, d^5 \; \Pi(\, \Psi(\eta_{\mathsf{max}}) \,)}_{\mathsf{const}} \; \Omega^3 = \mathsf{const} \, \Omega^3$$

the generator should follow the turbine in time-average

Feedback control law $P_{ m gen} = { m min}ig({\color{red} a \, \Omega^b}, \, P_{ m gen}^{ m rated} \, ig)$ $T_{ m gen} = { m min}igg({\color{red} a \, \Omega^{b-1}}, \, {\color{red} rac{P_{ m gen}}{ m gen}} \, igg)$

• The control law is clipped at generator rated power



Generator feedback control law

- Constants "a" and "b" are functions of the WEC hydrodynamics and turbine geometry but not of sea-state
- Usually 2.4 < b < 3.6
- To compute "a" and "b" we use a constant rotational speed model

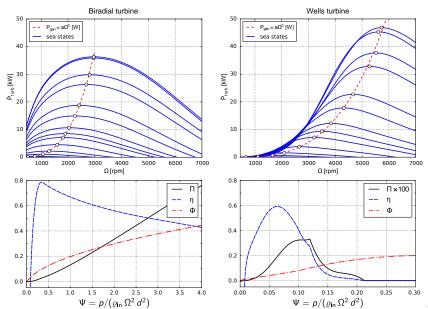
$$\dot{\Omega}^* = 0 \ \ \, \Rightarrow \ \ \, \text{very large inertia}$$

for the Mutriku wave climate (14 sea states)

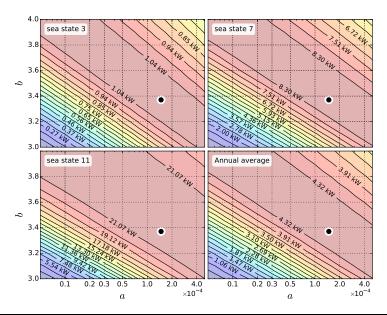
• Further details can be found in [1, 2]



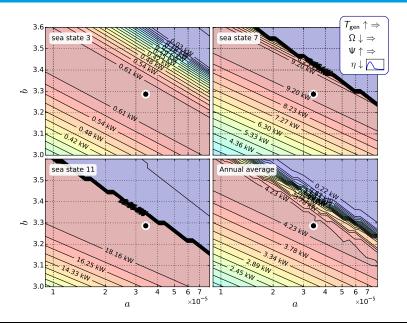
Generator feedback control law



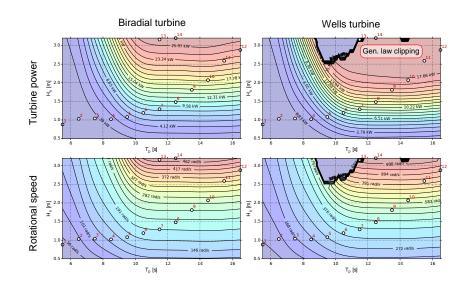
Biradial turbine power output sensitivity to "a" and "b"



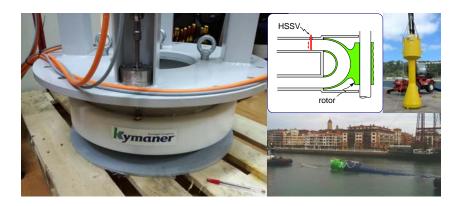
Wells turbine power output sensitivity to "a" and "b"



Comparison time-averaged turbine output power and rotational speed



PART 2 - Optimal control of the turbine High-Speed Stop Valve (Latching)



- The HSSV valve of a biradial turbine with 0.5 m rotor diameter
 - ► HSSV movie
- Latching with the HSSV will be tested at Mutriku, June-July 2017 (OPERA)
- HSSV will operate as safety valve in Oceantec MARMOK-A-5, October 2017 (OPERA). Buoy: 5 m diameter, 42 m in length and 80 tonnes weight.

Latching control and air compressibility

- OWCs vs oscillating body WECs
 - Latching on OWCs not so effective as in the case of an oscillating-body WECs
 - air compressibility has a spring effect
 - closing the HSSV does not stop the relative motion between the OWC and buoy
 - Decreases the forces resulting from latching in comparison with solid-body breaking
 - the valve surface area subject to the chamber pressure is a small fraction of the area of the OWC free surface
 - no impact forces air compressibility spring effect
 - Removes the constraint of latching having to coincide with an instant of zero relative velocity between the floater and the OWC

Changes to the mathematical model of Mutriku

Spar-buoy OWC - two body system constrained to oscillate in heave

Buoy:
$$(m_1 + A_{11}^{\infty}) \ddot{x}_1 + A_{12}^{\infty} \ddot{x}_2 = -\rho_w g S_1 x_1 - R_{11} - R_{12} + F_{d1} + \rho_{atm} S_2 p^*$$

OWC:
$$A_{21}^{\infty}\ddot{x}_1 + (m_2 + A_{22}^{\infty})\ddot{x}_2 = \underbrace{-\rho_w g S_2 x_2}_{\text{buoyancy}}\underbrace{-R_{21} - R_{22}}_{\text{radiation}}\underbrace{+F_{d2}}_{\text{diffraction}}\underbrace{-\rho_{\text{atm}} S_2 p^*}_{\text{air chamber}}$$

• HSSV simulation requires a new definition of Ψ

$$\tilde{\Psi} = u \Psi = u \frac{p^* p_{\text{at}}}{\rho_{\text{in}} \Omega^2 d^2}$$

where $u \in \{0,1\}$ is the **discrete** control (close/open)

System of equations written as a 1st-order ODE

$$\dot{\mathbf{x}} = \mathbf{f}(t, \mathbf{x}, u) \tag{1}$$

x are called the states



Optimal control and the Pontriagyn Maximum Principle

Optimal control

Maximization of a performance index based on the control u of the HSSV

$$\max \left(\int_0^T \mathcal{L}\left(\mathbf{x}, \mathbf{u}\right) dt \right)$$

- $\bullet \ \, \mathsf{Example:} \ \, \mathcal{L}\left(\mathbf{x},\!\mathbf{u}\right) = \frac{P_{\mathsf{turb}}(t,\!\mathbf{x},\!u)}{\rho_{\mathsf{atm}}\Omega_{\mathsf{max}}^{3}d^{5}}$
- Using the Pontriagyn Maximum Principle (PMP) the optimal problem is recast as

$$\max \left(\underbrace{\int_{0}^{T} \mathcal{L}(\mathbf{x}, u) \, \mathrm{d}t}_{\text{performance index}} - \underbrace{\int_{0}^{T_{f}} \boldsymbol{\lambda}^{T} \left(\dot{\mathbf{x}} - \mathbf{f}(t, \mathbf{x}, u)\right) \, \mathrm{d}t}_{\text{constrained to system dynamics}} \right)$$

• λ' s are called adjoint variables



Solution of the Pontriagyn Maximum Principle

- The PMP shows that:
 - ullet along the **optimal control path** (\mathbf{x},u) the Hamiltonian function $\mathcal H$

$$\mathcal{H}(t,\mathbf{x},\mathbf{u},\boldsymbol{\lambda}) = \boldsymbol{\lambda}^{\mathsf{T}} \mathbf{f}(t,\mathbf{x},\mathbf{u}) + \mathcal{L}(\mathbf{x},u),$$

is maximum for the optimal input u subjected to:

$$\begin{array}{c} \dot{\mathbf{x}} = \mathbf{f}(t, \mathbf{x}, u) \\ \mathbf{x}(0) = \mathbf{x}_0 \end{array} \right\} \Rightarrow \text{forward solution}$$

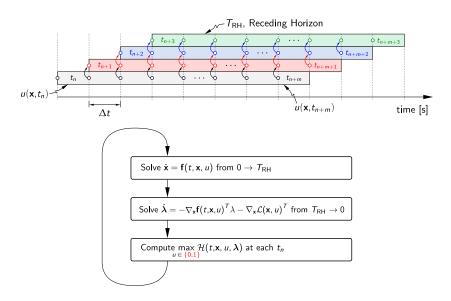
$$\left. \begin{array}{l} \dot{\boldsymbol{\lambda}} = -\nabla_{\mathbf{x}}\mathbf{f}(t,\mathbf{x},u)^{\mathsf{T}}\boldsymbol{\lambda} - \nabla_{\mathbf{x}}\mathcal{L}(\mathbf{x},u)^{\mathsf{T}} \\ \boldsymbol{\lambda}(T_f) = 0 \end{array} \right\} \Rightarrow \mathsf{backward} \; \mathsf{solution}$$

for $t \in [0, T_f]$

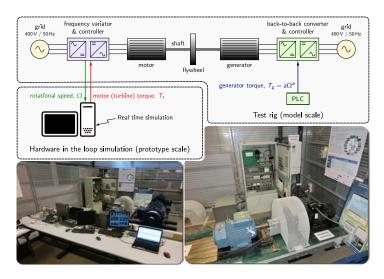
- non-causal control $F_d(t)$ required to compute **f** for $t \in [0, T_f]$
- For complete details see the two classical books of Luenberger [5] and Bryson & Ho [6]



Receding Horizon control in a real-time framework

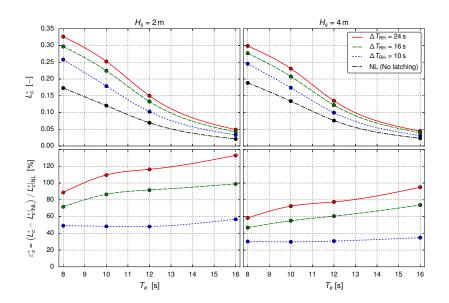


Hardware-in-the-loop tests at Tecnalia test rig

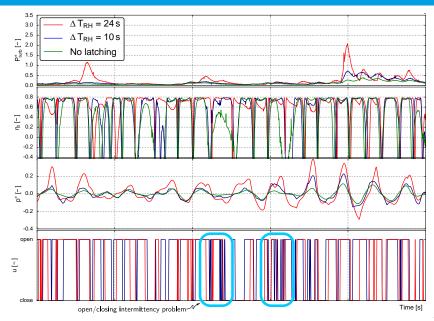


See [3, 4] for further details

Hardware-in-the-loop tests at Tecnalia test rig



Hardware-in-the-loop tests at Tecnalia test rig

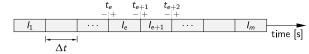


Lessons learned from the Tecnalia hardware-in-the-loop tests

- To improve stability, all state variables must have the same order of magnitude
- The algorithm sometimes opens/closes the HSSV intermittently during short periods (less than 1.5 seconds)
- The discrete nature of the numerical solution of the PMP problem implies a 2nd-order accuracy
- A new challenge in the HSSV control
 - The sub-optimal problem of open/closing the HSSV only at the end of each (non-infinitesimal) time step, $\Delta t = 0.1 \sim 0.2 \, s$
 - No solution found in the searched bibliography!
 - We start the quest for a new solution method...



The computational domain is discretized in small time-elements



where

- State, adjoint and control variables approximated by a set of Legendre polynomials, $p_j(t)$ such that $x = \sum_i p_j(t) \tilde{x}_j$ in each time element
- The numerical solution is continuous within the time elements and allowed to be discontinuous across element boundaries
- \bullet Solution of the sub-optimal control problem compute u that maximizes the Hamiltonian ${\mathcal H}$ in a integral sense

$$\max_{\mathbf{u} \in \{0,1\}} \, \int_{t_{\mathbf{e}}}^{t_{\mathbf{e}+1}} \mathcal{H}(t,\!\mathbf{x},\mathbf{u},\boldsymbol{\lambda}) \, \mathrm{d}t$$



In the DG Finite Element Method, the original problem

$$\dot{\mathbf{x}} - \mathbf{f}(t, \mathbf{x}, \mathbf{u}) = 0$$

is replaced by a weak formulation

$$\int_{I_e} v_h \underbrace{\left(\frac{\mathrm{d} x_h}{\mathrm{d} t} - f\right)}_{\text{state equations}} \, \mathrm{d} t + \underbrace{v_h(t_e) \, \left[x_h(t_e^+) - x(t_e^-)\right]}_{\text{weakly enforced BC}} = 0$$

 v_h is the so-called test function in the FEM framework

• Applying an affine transformation from $t \in I_e$ to $\tau \in [-1,1]$

$$\int_{-1}^{1} \hat{v}_{h} \frac{\mathrm{d}\hat{x}_{h}}{\mathrm{d}\tau} \,\mathrm{d}\tau + \hat{v}_{h}(-1) \,\hat{x}_{h}(-1^{+}) = \frac{\Delta t}{2} \int_{-1}^{1} \hat{v}_{h} \,\hat{f} \,\mathrm{d}\tau + \hat{v}_{h}(-1) \,\hat{x}(-1^{-}) \tag{2}$$

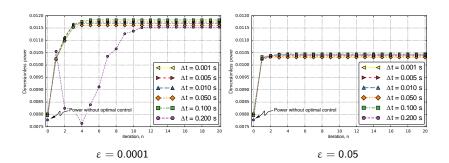
we write (2) as linear system $A \hat{x}_h = b(t, \mathbf{x}, u)$ in each element

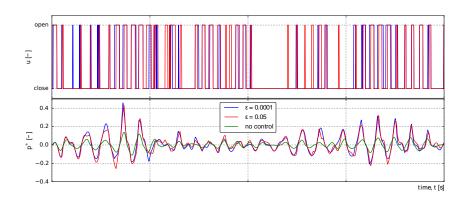


• To improve the convergence rate, a regularization term was introduced

$$\mathcal{L}(\mathbf{x}, u) = \frac{1}{T} \int_{0}^{T} \left(\frac{P_{\mathrm{turb}}(t, \mathbf{x}, u)}{\rho_{\mathrm{atm}} \Omega_{\mathrm{max}}^{3} d^{5}} + \varepsilon \left(1 - u \right)^{2} \right) \, \mathrm{d}t$$

 ε is a small constant





Major conclusions

- Receding horizon optimal latching algorithm can be used to improve the OWC spar-buoy capture width
 - Receding horizon time interval between 10 to 24 s
- The Discontinuous-Galerkin solves the HSSV open/closing intermittency problem of the discrete optimal control
- Probably the greatest advantage of the OWC technology
 - Simple control of the available power to PTO system by using a HSSV
 - HSSV can be used for latching and control of the available power to the turbine and generator
- The proposed Discontinuous-Galerkin method is an efficient alternative to the well know Pseudo-Spectral Methods
 - High-order accuracy, mesh and polynomial (h-p) refinement, local nature of the solution, simple parallelization (real-time applications)

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